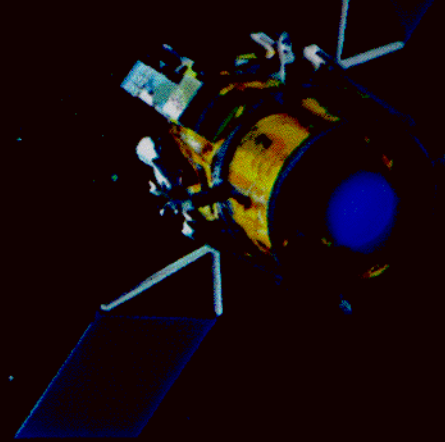
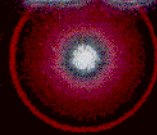
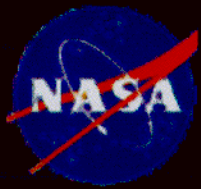


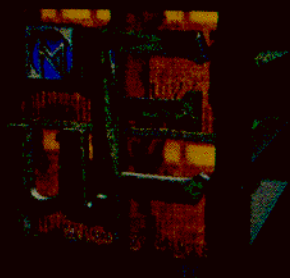
*New Millennium Program*  
*Deep Space 1*  
*Micro-Electronics Systems*  
*Technologies*

**Deep Space One**





# Micro-Electronic Systems Integrated Product Development Team



The Team  
&  
Our Technologies



# The Micro-Electronics Systems Integrated Product Development Team



National Aeronautics and  
Space Administration



Jet Propulsion  
Laboratory



Lewis Research Center



Sandia  
National Labs



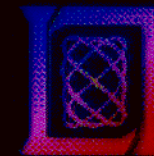
Air Force/  
Phillips Laboratory



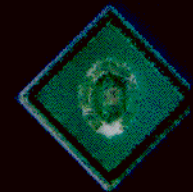
TRW



Lockheed Martin



MIT/  
Lincoln Labs



Johns Hopkins University/  
Applied Physics Lab



University of  
Southern California



Hughes Telecommunications  
& Space



Georgia Institute  
of Technology



Honeywell



Optivision  
Corporation



Boeing



Goddard Space  
Flight Center



Irvine Sensors  
Corporation



Space Computer  
Corporation

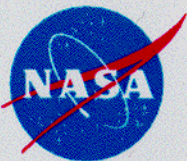




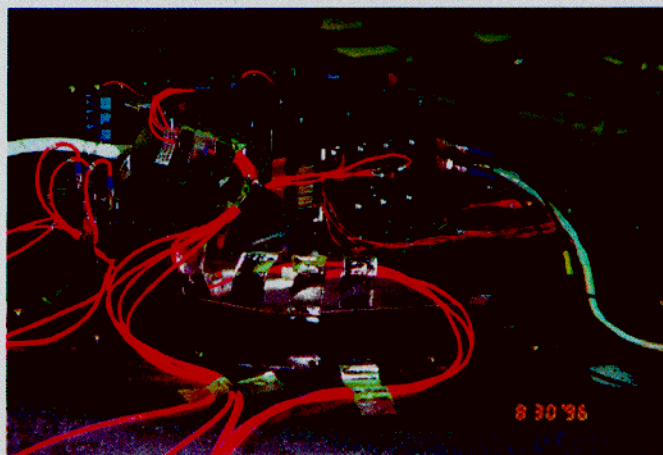
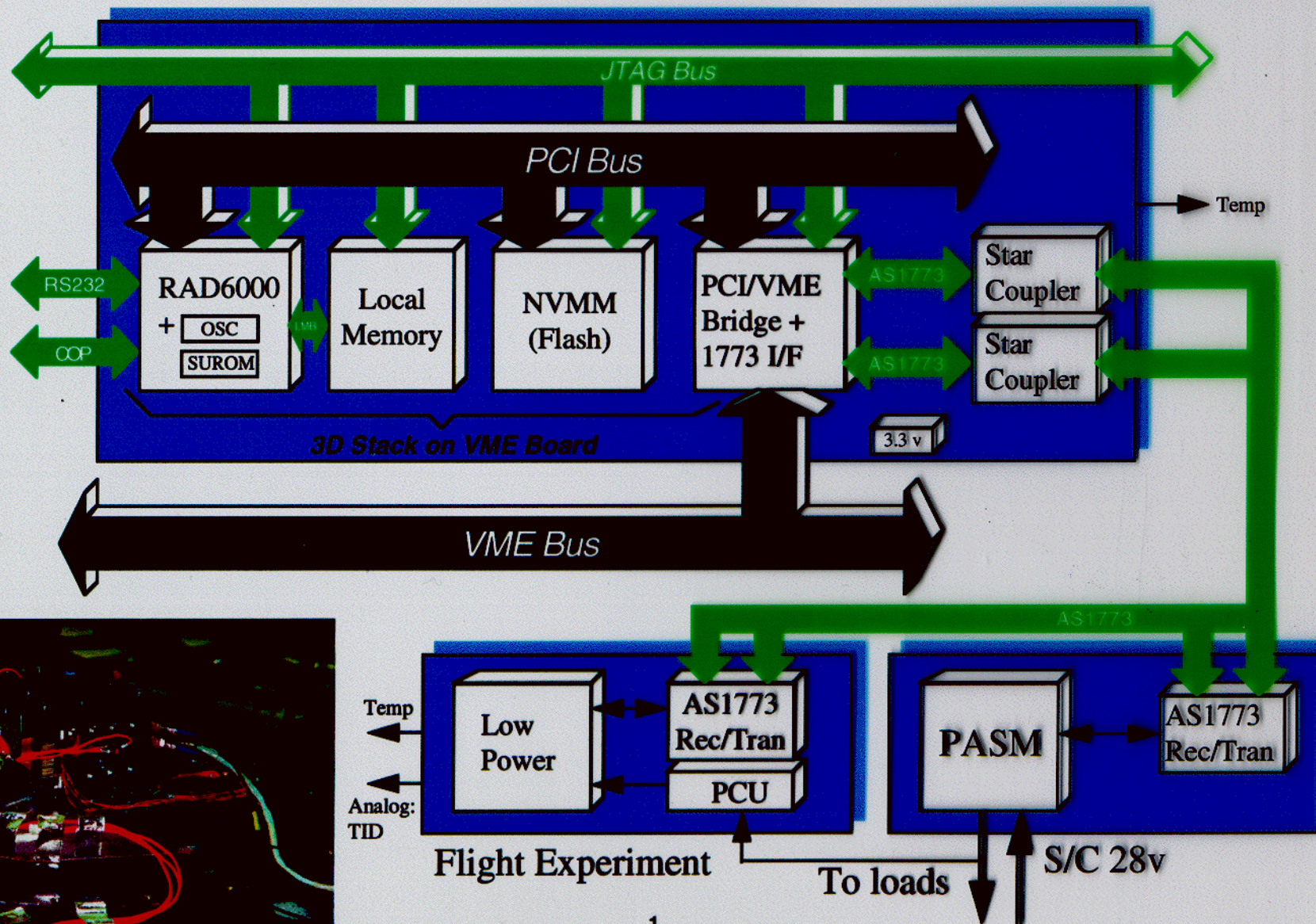
# Current Technologies

- 3D Stack Space Computer
- Mass Storage
- Low Power Electronics
- Power Actuation and Switching Module
- Distributed Reliable Computing





# Example of the NMP/DS1 System Architecture







# 3-D Packaging

- **Capabilities**

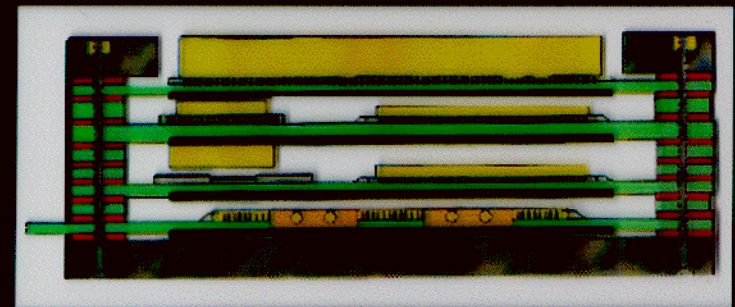
- Stack of 4 MCMs (Processor, Memory, I/O, SSR)
- 5.8" x 4.0" x 2.1"
- 360 I/O Pads @ 32-mil Pitch
- MCM Bonded to Substrate Holders. Substrate Holders Stacked and Connected by Elastomeric Elements
- Re-workable Without Special Process
- Design for Testability with Boundary Scan

- **Partnership**

- Space Computer Corporation

- **Benefits**

- Step Toward Direct MCM Stacking





# 3-D Stack, Top and Side Views

Printed Circuit Boards  
0.090" 0.150" 0.090

4.000  
inches

4.000  
inches

5.800  
inches

TRW  
Solid-State Recorder  
Module

Threaded Insert

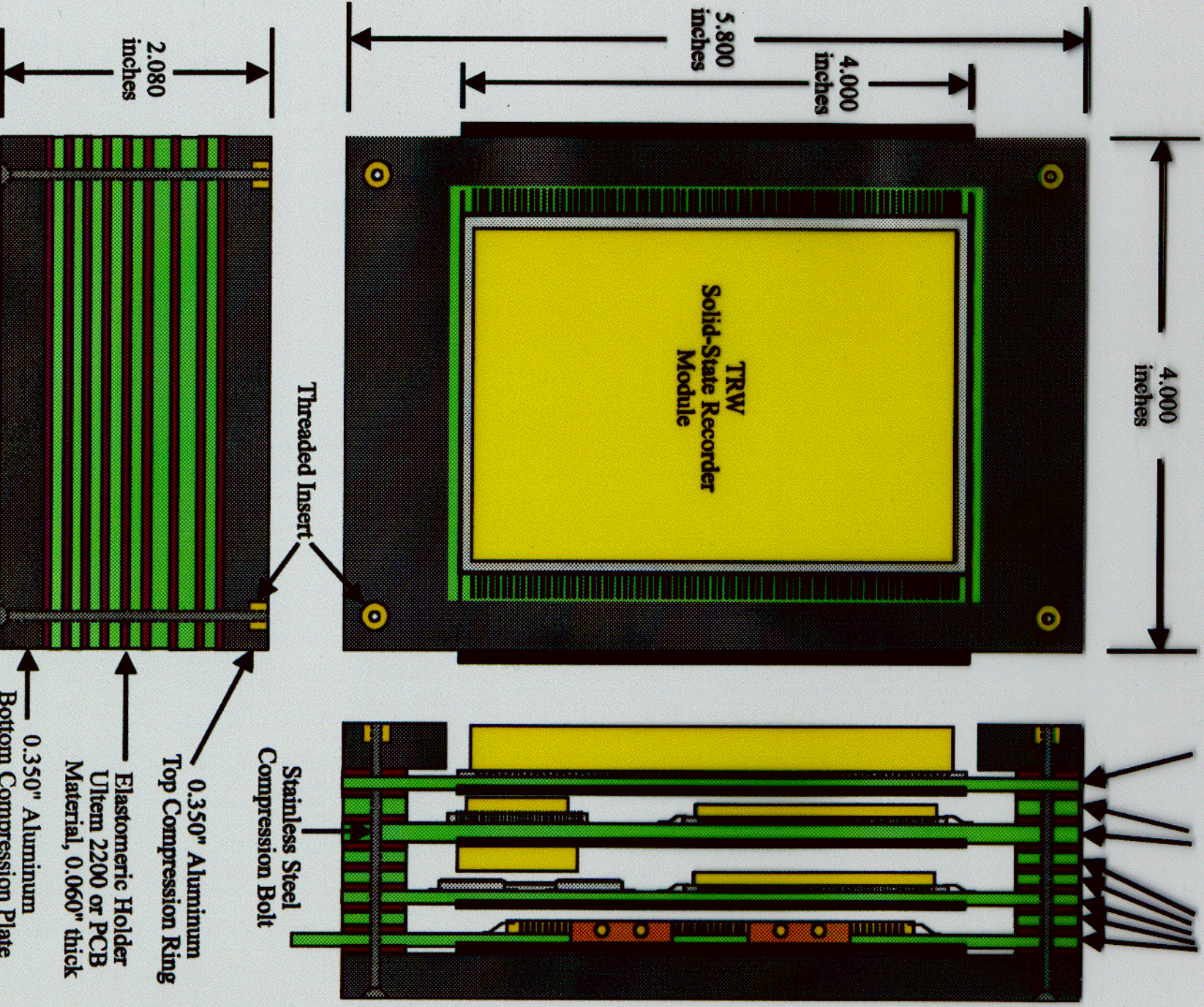
Stainless Steel  
Compression Bolt

0.350" Aluminum  
Top Compression Ring

Elastomeric Holder  
Ultem 2200 or PCB  
Material, 0.060" thick

0.350" Aluminum  
Bottom Compression Plate

2.080  
inches



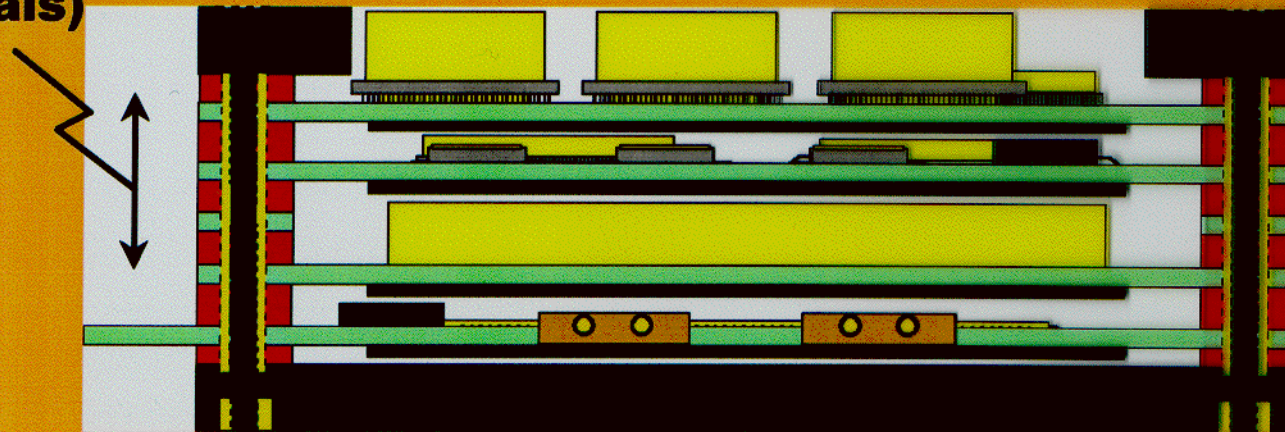
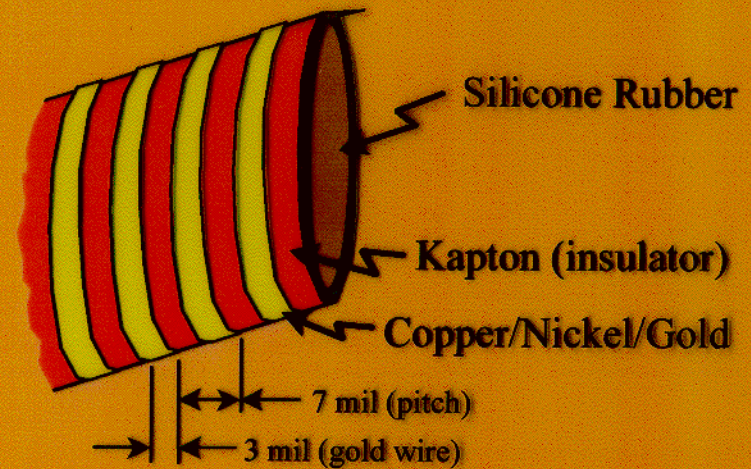


## Microelectronics 3D Stack

# Elastomeric Interconnect

**The elastomer provides a highly dense interconnection mechanism between “slices” in the stack**

**Each interconnect side of the stack has 156 signals (312 total interconnect signals)**







# 3D Stack Space Computer

- 5.8" x 4.0" x 2.1" Form Factor
- 3 Dimensional Stacking of Slices
- 33 MHz Rad-6000
  - 32 Bit RISC Processor
  - Commercial PCI Interface
  - Commercial OS: VxWorks
- 320 MB DRAM
  - 100x Reduction of Mass & Volume Relative to PCB Technology
- 1773 Interface
- Point of Contact: Dwight Geer, JPL

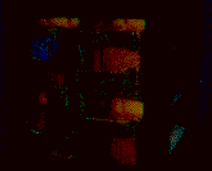


**Memory Slice**



**Processor Slice**





# Mass Storage

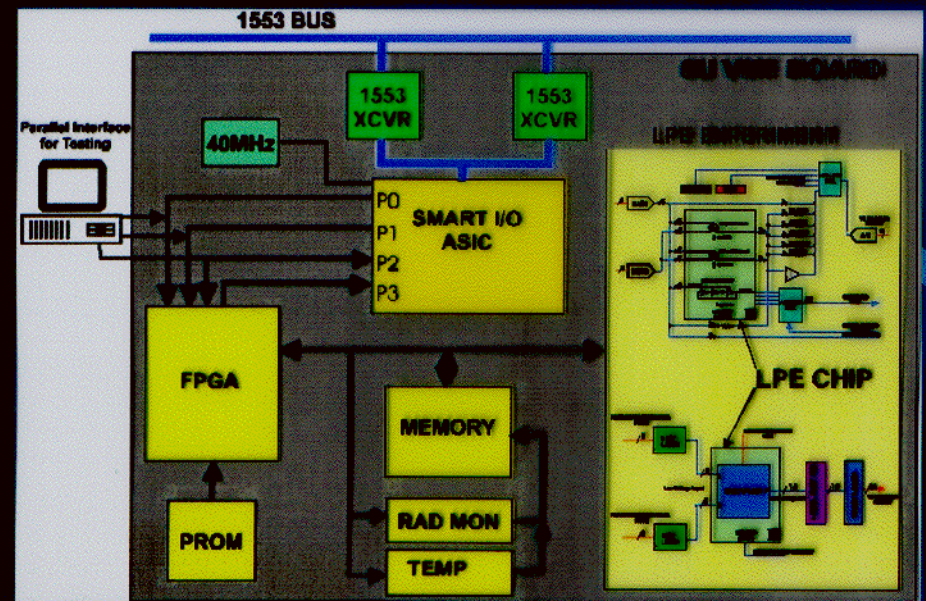
- Non-Volatile Memory Modules
  - 1 Gigabit Storage Per Slice
  - Can Have Multiple Slices
  - All Solid State, No Moving Parts
- Point of Contact: Dwight Geer, JPL





# Low Power Electronics

- .25  $\mu\text{m}$  Silicon-on-Insulator CMOS
  - Extremely Low Power
  - Inherently Rad-hard
  - Capable of Extremely High Levels of Integration
  - 0.18  $\mu\text{m}$  CMOS/SOI is being developed
- Point of Contact:
  - Eric Holmberg, JPL

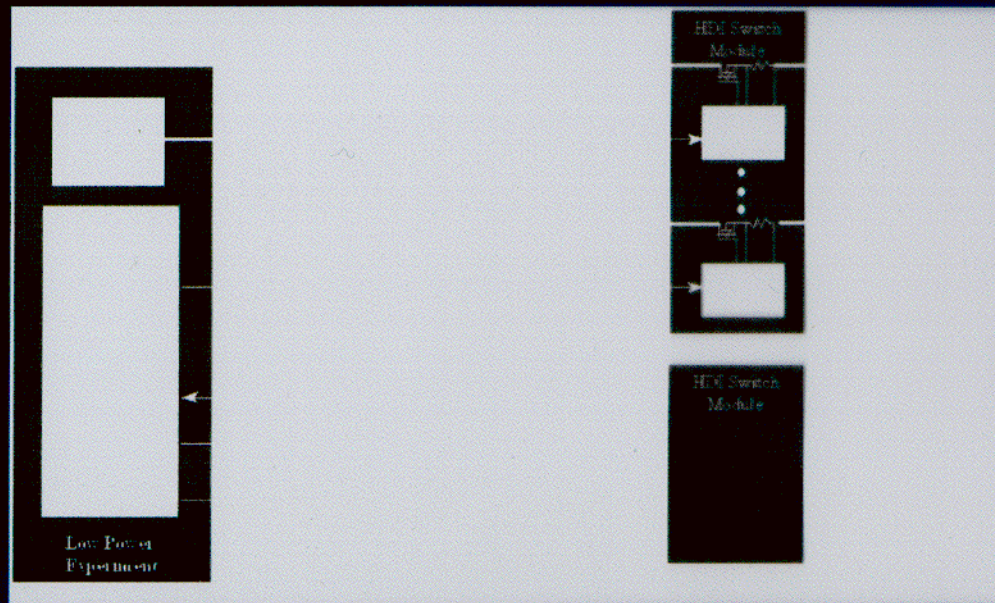






# Power Actuation and Switching Module

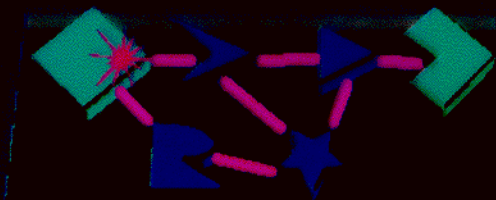
- High Voltage, Mixed Signal ASICs
- Uses HDI Technology
- 8 Power Switches Outputs
- Point of Contact: Greg Carr, JPL





# Distributed Reliable Computing

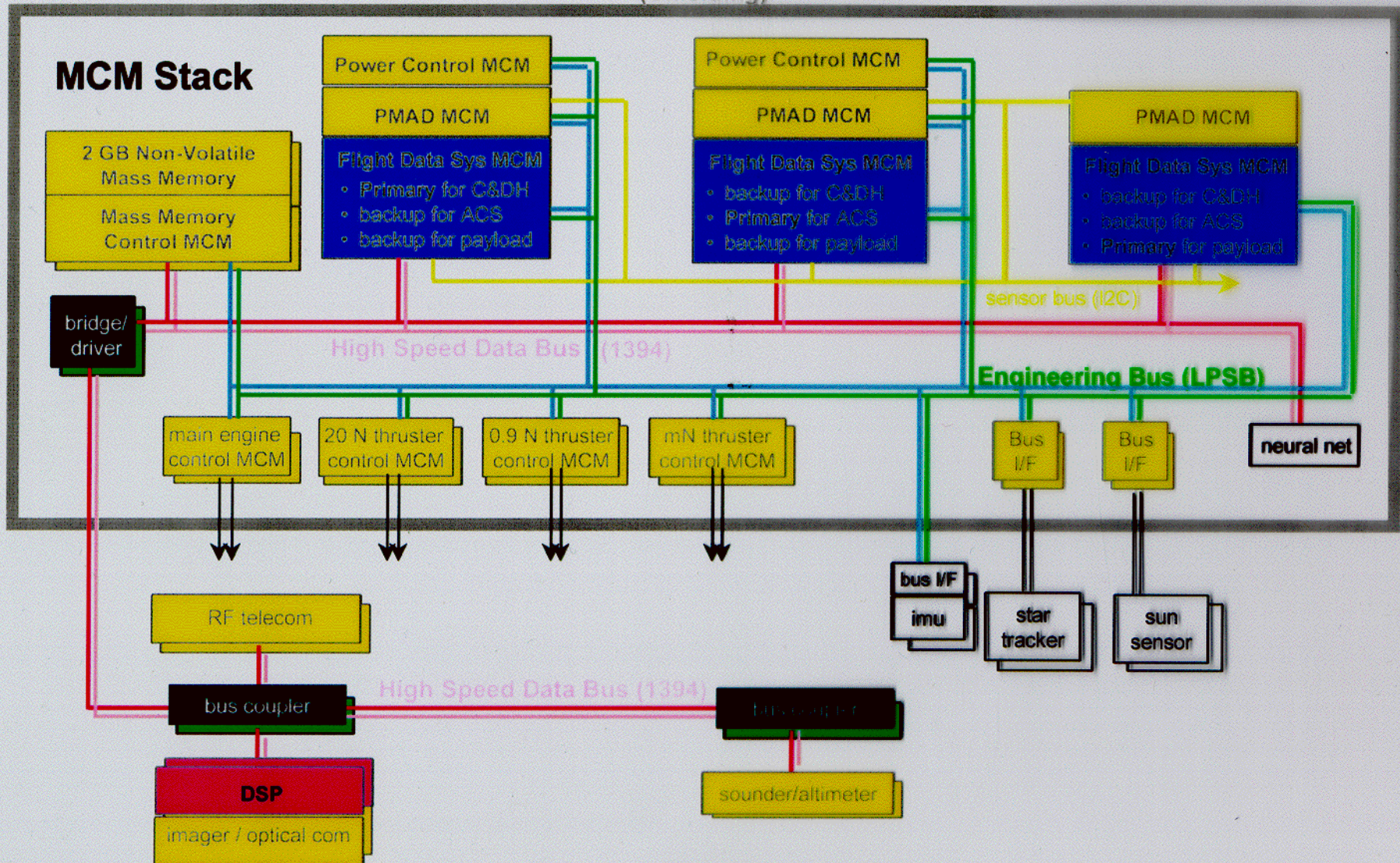
- Commercial Off The Shelf Hardware
- Asynchronous, Message Based Application Software
- *Software Implemented Fault Tolerance* Inserted Into Message Passing System
- *Integrated I/O Architecture* Significantly Reduces Software Complexity and Performance Requirements
- *Heirarchical, Reliable Network/bus Architecture* Supporting Standard Network Protocols
- Point of Contact: David Smythe, JPL



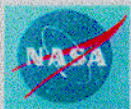


# X2000 Avionics Architecture

(Shielding)

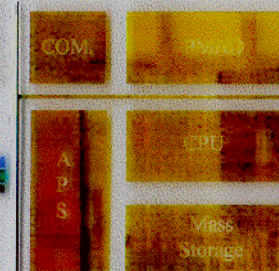






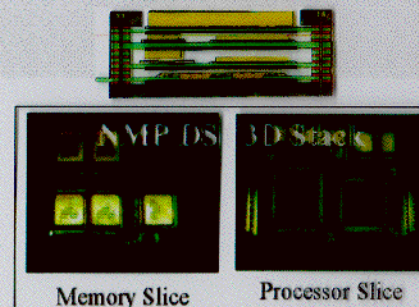
# Evolution of Computers for Deep Space

**\*\*>2000X<sub>rFOM</sub>**



- 100s of MIPS
- Functionality:**
- \*Command and Data Handling
  - \*Attitude Determination and Control
  - \*Mass Storage
  - \*PMAD
  - \*Sensor electronics, mixed signal, DSP
  - \*Telecom
  - \*GN&C sensors and MEMS
  - \*Evolvable system
  - \*Reconfigurable Hardware
  - \*Major use of IP

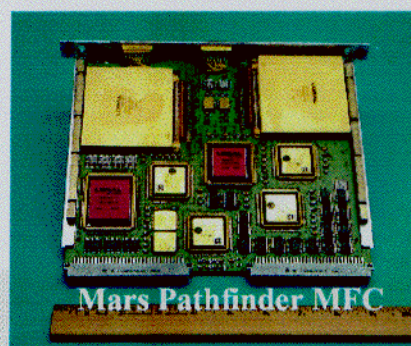
**\*\*117X<sub>rFOM</sub>**



: RAD6000  
RISC Flight Computer  
2.5 to 55MIPS, very high  
perf. PCI and 1773 I/O,  
low power electronics

- Functionality:**
- \*Command and Data Handling
  - \*Attitude Determination and Control
  - \*Mass Storage (nonvolatile)
  - \*Power Management & Distribution (PMAD)

**\*\*20X<sub>rFOM</sub>**



RAD6000 RISC  
2.5 to 22  
MIPS

**Functionality:**

- \*Command and Data Handling
- \*Attitude Determination and Control

*Goal: Breakthrough Increase in  
\*Functional Density and \*\*Performance  
While Decreasing Cost, Power and Mass*

$$**rFOM = \frac{\text{Delay} \times \text{Power}}{\text{Density}}$$

Relative Figure of Merit at IC Level

1750 A  
: 1.25 MIPS

**Functionality:**

- \*Command and Data Handling

**\*\*1X<sub>rFOM</sub>**

